

MODAL RESPONSE CONTRIBUTIONS

Modal Expansion of Excitation Vector $\mathbf{p}(t) = \mathbf{s}p(t)$

A common loading case is the one in which the applied forces $p_j(t)$ have the **same time variation** $p(t)$, and their **spatial distribution** is defined by \mathbf{s} , **independent of time**.

$$\mathbf{p}(t) = \mathbf{s}p(t)$$

We **expand** the vector \mathbf{s} in its modal components as:

$$\mathbf{s} = \sum_{r=1}^N \mathbf{s}_r = \sum_{r=1}^N \Gamma_r \mathbf{m} \boldsymbol{\phi}_r$$

Pre-multiplying by $\boldsymbol{\phi}_n^T$ and utilizing the orthogonality property of the modes, we obtain:

$$\Gamma_n = \frac{\boldsymbol{\phi}_n^T \mathbf{s}}{M_n} \quad \text{Modal Participation Factor}$$

It is evident that Γ_n is **not independent** of how the mode is normalized.

The contribution of the n^{th} mode to \mathbf{s} is:

$$\mathbf{s}_n = \Gamma_n \mathbf{m} \boldsymbol{\phi}_n$$

which is **independent** of how the modes are normalized.

The inertia forces, associated with the n^{th} mode, are:

$$(\mathbf{f}_I)_n = -\mathbf{m} \ddot{u}_n(t) = -\mathbf{m} \boldsymbol{\phi}_n \ddot{q}_n(t)$$

Clearly, the spatial distribution of the inertia forces $(\mathbf{f}_I)_n$ associated with the n^{th} mode, is the same as that of \mathbf{s}_n .

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The expansion,

$$\mathbf{s} = \sum_{r=1}^N \mathbf{s}_r = \sum_{r=1}^N \Gamma_r \mathbf{m} \boldsymbol{\phi}_r$$

has two useful properties:

- (1) The force vector $\mathbf{s}_n p(t)$ produces response only in the n^{th} mode but no response in any other mode.

Proof:

Generalized force associated with $\mathbf{s}_n p(t)$:

$$P_r(t) = \boldsymbol{\phi}_r^T \mathbf{s}_n p(t) = \Gamma_r \underbrace{(\boldsymbol{\phi}_r^T \mathbf{m} \boldsymbol{\phi}_n)}_{M_n \delta_{nr}} p(t) = \Gamma_r M_n \delta_{nr} p(t)$$

[δ_{nr} : Kronecker delta]

- (2) The dynamic response of the n^{th} mode is entirely due to the partial force vector $\mathbf{s}_n p(t)$.

Proof:

Generalized force associated with $\mathbf{s} p(t)$:

$$P_n(t) = \boldsymbol{\phi}_n^T \mathbf{s} p(t) = \sum_{r=1}^N \Gamma_r (\boldsymbol{\phi}_n^T \mathbf{m} \boldsymbol{\phi}_r) p(t) = \Gamma_n M_n p(t)$$

Evidently, the n^{th} mode generalized force associated with the complete force vector $\mathbf{s} p(t)$ is the same as the n^{th} mode generalized force associated with the partial force vectors $\mathbf{s}_n p(t)$.

Modal Analysis for $\mathbf{p}(t) = \mathbf{s}p(t)$

We have derived N uncoupled equations:

$$\ddot{q}_n(t) + 2\xi_n\omega_n\dot{q}_n(t) + \omega_n^2q_n(t) = \frac{P_n(t)}{M_n} \quad (n = 1, 2, \dots, N)$$

$$\text{where: } P_n(t) = \boldsymbol{\Phi}_n^T \mathbf{p}(t) = \boldsymbol{\Phi}_n^T \mathbf{s}p(t) = \Gamma_n M_n p(t)$$

Therefore:

$$\ddot{q}_n(t) + 2\xi_n\omega_n\dot{q}_n(t) + \omega_n^2q_n(t) = \Gamma_n p(t)$$

We can write the solution $q_n(t)$ as follows:

$$\boxed{\begin{aligned} \ddot{D}_n(t) + 2\xi_n\omega_n\dot{D}_n(t) + \omega_n^2D_n(t) &= p(t) \\ \text{and } q_n(t) &= \Gamma_n D_n(t) \end{aligned}}$$

Contribution of the n^{th} mode to $\mathbf{u}(t)$:

$$\boxed{\mathbf{u}_n(t) = \boldsymbol{\Phi}_n q_n(t) = \Gamma_n \boldsymbol{\Phi}_n D_n(t)}$$

Equivalent static forces $\mathbf{f}_{Sn}(t)$:

$$\begin{aligned} \mathbf{f}_{Sn}(t) = \mathbf{k}\mathbf{u}_n(t) &= \Gamma_n \underbrace{\mathbf{k}\boldsymbol{\Phi}_n}_{\omega_n^2 \mathbf{m}\boldsymbol{\Phi}_n} D_n(t) = \Gamma_n \underbrace{\mathbf{m}\boldsymbol{\Phi}_n}_{\mathbf{s}_n} [\omega_n^2 D_n(t)] \\ \Rightarrow \boxed{\mathbf{f}_{Sn}(t) &= \mathbf{s}_n [\omega_n^2 D_n(t)]} \end{aligned}$$

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Equivalent Static Forces $\mathbf{f}_{Sn}(t)$:

$$\mathbf{f}_{Sn}(t) = \mathbf{s}_n[\omega_n^2 D_n(t)]$$

The n^{th} mode contribution $r_n(t)$ to any response quantity $r(t)$ is determined by static analysis of the structure subjected to forces $\mathbf{f}_{Sn}(t)$.

Let:

r_n^{st}
modal static response = the static value of response quantity r due to external forces \mathbf{S}_n

Then:

$$r(t) = \sum_{n=1}^N r_n(t) = \sum_{n=1}^N r_n^{st}[\omega_n^2 D_n(t)]$$

The modal analysis procedure presented above provides a basis for identifying and understanding the factors that influence the relative modal contributions to the response.

NOTE:

Although we loosely refer to \mathbf{s}_n as forces, they are dimensionless because $p(t)$ has units of force.

Thus, r_n^{st} does not have the same units as r , but equation $r_n(t) = r_n^{st}[\omega_n^2 D_n(t)]$ gives the correct units for r_n and equation $r(t) = \sum_{n=1}^N r_n(t)$ for r .

Modal Contribution Factors

Let:

r^{st} = the static value of response quantity r due to external forces \mathbf{s}

Contribution of the n^{th} mode to response quantity r :

$$\begin{aligned} r_n(t) &= r_n^{st} [\omega_n^2 D_n(t)] = r^{st} \left(\frac{r_n^{st}}{r^{st}} \right) [\omega_n^2 D_n(t)] \\ &= r^{st} \bar{r}_n [\omega_n^2 D_n(t)] \end{aligned}$$

where: $\bar{r}_n = \left(\frac{r_n^{st}}{r^{st}} \right)$ n^{th} **modal contribution factor**

Properties of modal contribution factors \bar{r}_n :

- (1) They are **dimensionless**;
- (2) They are **independent of how modes are normalized**;
- (3) **The sum** of the modal contribution factors over all modes is unity, that is:

$$\sum_{n=1}^N \bar{r}_n = 1$$

Property #3 follows by the following reasoning:

$$\mathbf{s} = \sum_{n=1}^N \mathbf{s}_n \Rightarrow r^{st} = \sum_{n=1}^N r_n^{st} \Rightarrow 1 = \sum_{n=1}^N \underbrace{\left(\frac{r_n^{st}}{r^{st}} \right)}_{\bar{r}_n}$$

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MODAL ANALYSIS: SUMMARY

Dynamic response of an MODF system with classical damping to external forces $\mathbf{p}(t)$:

- (1) Define the **structural properties**:
 - (a) Determine the **mass matrix** \mathbf{m} and **stiffness matrix** \mathbf{k} ;
 - (b) Estimate the modal damping ratios ξ_n .

- (2) **Determine the natural frequencies** ω_n and **modes** $\boldsymbol{\phi}_n$;

[i.e., solve the **eigenvalue problem**: $(\mathbf{k} - \omega^2\mathbf{m})\boldsymbol{\phi} = \mathbf{0}$]

- (3) **Compute the response in each mode** by following the steps:
 - (a) Set up the following N **uncoupled equations** ($n = 1, 2, \dots, N$):

$$\begin{aligned} M_n \ddot{q}_n(t) + C_n \dot{q}_n(t) + K_n q_n(t) &= P_n(t) \\ \Rightarrow \ddot{q}_n(t) + 2\xi_n \omega_n \dot{q}_n(t) + \omega_n^2 q_n(t) &= \frac{P_n(t)}{M_n} \end{aligned}$$

and solve for $q_n(t)$.

- (b) Compute the **nodal displacements** $\mathbf{u}_n(t) = \boldsymbol{\phi}_n q_n(t)$
- (c) Compute the **element forces** associated with $u_n(t)$ by implementing the *equivalent static force method*.

$$\mathbf{f}_{Sn}(t) = \mathbf{k}\mathbf{u}_n(t) \Rightarrow \mathbf{f}_{Sn}(t) = \omega_n^2 \mathbf{m}\boldsymbol{\phi}_n q_n(t) \quad (n = 1, 2, \dots, N)$$

- (4) Combine the contributions of all the modes to determine the response:

$$\begin{aligned} \left. \begin{array}{l} \text{nodal} \\ \text{displacements} \end{array} \right\} \mathbf{u}(t) &= \sum_{n=1}^N \mathbf{u}_n(t) = \sum_{n=1}^N \boldsymbol{\phi}_n q_n(t) \\ \text{e.g.} \quad \left. \begin{array}{l} \text{element} \\ \text{forces} \end{array} \right\} r(t) &= \sum_{n=1}^N r_n(t) \end{aligned}$$

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MODAL ANALYSIS FOR $\mathbf{p}(t) = \mathbf{s}p(t)$

After we determine:

- the **structural properties** (i.e., \mathbf{m} & \mathbf{k} , ξ_n).
- the **vibration properties** (i.e., ω_n & Φ_n).

we expand the force distribution \mathbf{s} into its modal components \mathbf{s}_n , i.e.,

$$\mathbf{s} = \sum_{n=1}^N \mathbf{s}_n = \sum_{n=1}^N \{\Gamma_n \mathbf{m} \Phi_n\}$$

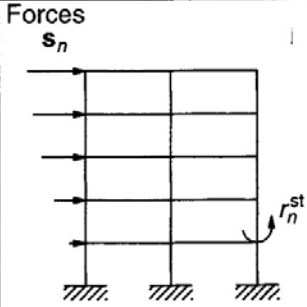
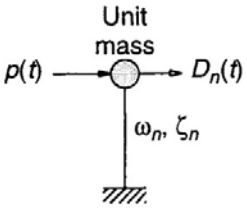
The contribution $r_n(t)$ of the n^{th} mode to the dynamic response is obtained by multiplying the results of two analyses:

- (1) *Static analysis* of the structure subjected to external forces \mathbf{s}_n ($n = 1, 2, \dots, N$).
- (2) *Dynamic analysis* of the n^{th} mode SDOF system excited by the force $p(t)$:

$$\begin{aligned} \ddot{D}_n(t) + 2\xi_n \omega_n \dot{D}_n(t) + \omega_n^2 D_n(t) &= p(t) \quad (n = 1, 2, \dots, N) \\ q_n(t) &= \Gamma_n D_n(t) \end{aligned}$$

Combining the modal responses gives the dynamic response of the structure.

Conceptual explanation of modal analysis

Mode	Static Analysis of Structure	Dynamic Analysis of SDF System	Modal Contribution to Dynamic Response
n	 <p>Forces s_n</p> <p>r_n^{st}</p>	 <p>Unit mass</p> <p>$p(t)$ $D_n(t)$</p> <p>ω_n, ζ_n</p>	$r_n(t) = r_n^{st} [\omega_n^2 D_n(t)]$

Modal Responses: $n = 1, 2, \dots, N$

$$r_n(t) = r_n^{st} [\omega_n^2 D_n(t)]$$

Total Response

$$r(t) = \sum_{n=1}^N r_n(t) = \sum_{n=1}^N r_n^{st} [\omega_n^2 D_n(t)]$$

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Modal Responses and Required Number of Modes

Let:

$$D_{no} \stackrel{\text{def}}{=} \max_t |D_n(t)|$$

Then, the corresponding value of $r_n(t)$ is:

$$r_{no} = \underbrace{r^{st} \bar{r}_n}_{r_n^{st}} \omega_n^2 D_{no}$$

Let:

$$R_{dn} = \frac{D_{no}}{(D_{n,st})_o} \quad \begin{array}{l} \text{Dynamic} \\ \text{Response Factor} \end{array}$$

where:

$$\begin{aligned} (D_{n,st})_o &= \max_t \underbrace{|D_{n,st}(t)|}_{\text{static response}} \\ &= \left(\frac{\max_t |p(t)|}{\omega_n^2} \right) \\ &= \left(\frac{p_o}{\omega_n^2} \right) \end{aligned}$$

The static response $D_{n,st}(t)$ is obtained from $\ddot{D}_n(t) + 2\xi_n \omega_n \dot{D}_n(t) + \omega_n^2 D_n(t) = p(t)$, by dropping the \ddot{D}_n & \dot{D}_n terms.

Therefore:

$$r_{no} = r^{st} \bar{r}_n p_o R_{dn}$$

The algebraic sign of r_{no} is the same as that of $r_n^{st} \stackrel{\text{def}}{=} r^{st} \bar{r}_n$ because R_{dn} is positive by definition.

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Peak Modal Response

$$r_{no} = r^{st} \bar{r}_n p_o R_{dn}$$

Peak modal response is the product of four quantities:

- (1) R_{dn} (=dimensionless) *dynamic response factor* for the n^{th} mode SDOF excited by force $p(t)$
- (2) p_o , the *peak value* of $p(t)$
- (3) \bar{r}_n (=dimensionless) *modal contribution factor* for response quantity r
- (4) r^{st} , the *static value* of r due to external forces \mathbf{s}

Clearly:

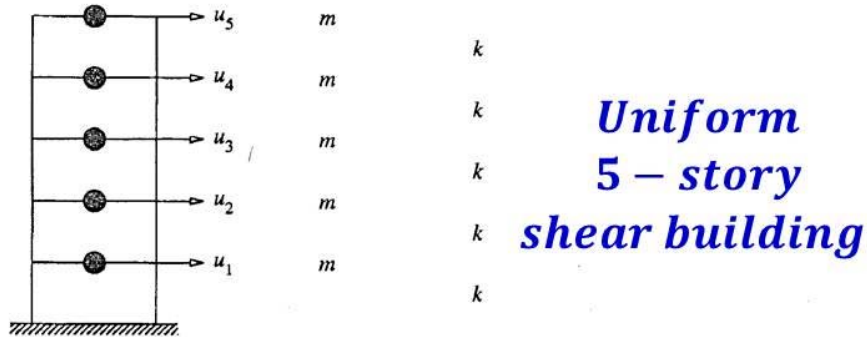
r^{st} & \bar{r}_n : depend on the **spatial distribution** \mathbf{s} of the applied forces but are independent of the **time variation** $p(t)$ of the applied forces.

R_{dn} : depends on $p(t)$, but is independent of \mathbf{s} .

\bar{r}_n & R_{dn} influence the relative response contributions of the various vibration modes, and hence the minimum number of modes that should be included in the dynamic analysis.

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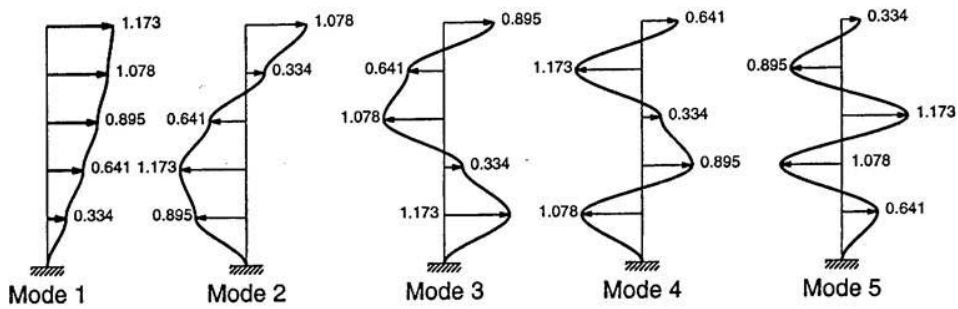
EXAMPLE:



The mass and stiffness matrices of the structure are

$$\mathbf{m} = m \begin{bmatrix} 1 & & & & \\ & 1 & & & \\ & & 1 & & \\ & & & 1 & \\ & & & & 1 \end{bmatrix} \quad \mathbf{k} = k \begin{bmatrix} 2 & -1 & & & \\ -1 & 2 & -1 & & \\ & -1 & 2 & -1 & \\ & & -1 & 2 & -1 \\ & & & -1 & 1 \end{bmatrix}$$

Natural vibration modes of five-story shear building



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The relative contributions of the various modes to the response,

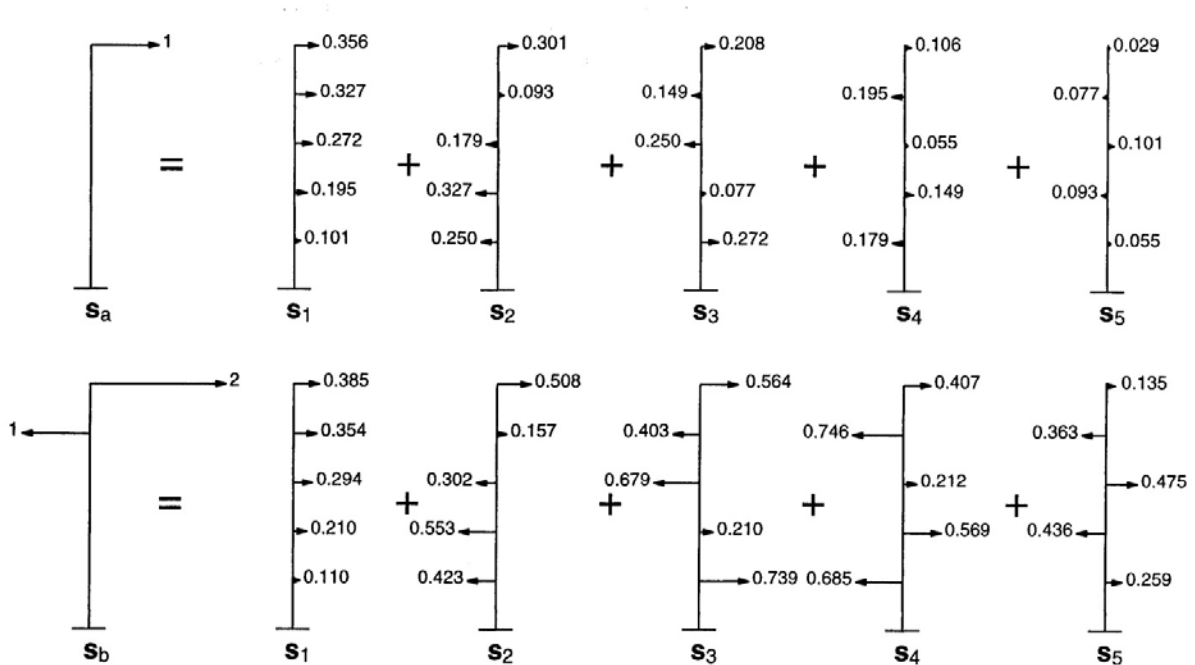
and

the number of modes that **should be included** in dynamic analysis

depend on the following **two influencing factors**:

- (1) The *modal contribution factor* \bar{r}_n , which depends on the **spatial distribution of forces**.
- (2) The *dynamic response factor* R_{dn} , which is controlled by the **time variation of forces**.

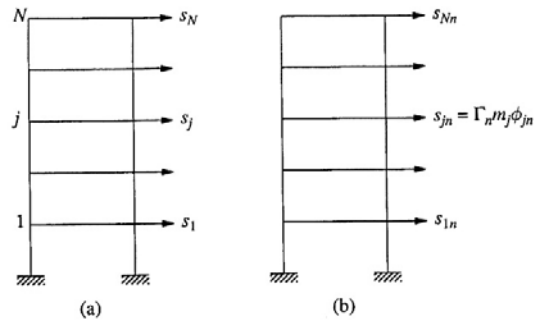
Modal expansion of excitation vectors \mathbf{s}_a and \mathbf{s}_b



Observation:

The contributions of the higher modes to \mathbf{s} are larger for \mathbf{s}_b than for \mathbf{s}_a , suggesting that these modes (*i.e.*, the higher modes) may contribute more to the response if the force distribution is \mathbf{s}_b than if it is \mathbf{s}_a .

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Determination of the *modal contribution factors* for:

- (i) The base shear V_b
- (ii) The roof (N th floor) displacement u_N

$$V_{bn}^{st} = \sum_{j=1}^N S_{jn} = \Gamma_n \sum_{j=1}^N m_j \phi_{jn} =$$

$$\mathbf{u}_n^{st} = \mathbf{k}^{-1} \mathbf{s}_n = \mathbf{k}^{-1} \Gamma_n \mathbf{m} \boldsymbol{\phi}_n = \frac{\Gamma_n}{\omega_n^2} \boldsymbol{\phi}_n$$

(Recall that: $\mathbf{k} \boldsymbol{\phi}_n = \omega_n^2 \mathbf{m} \boldsymbol{\phi}_n \Leftrightarrow \frac{1}{\omega_n^2} \boldsymbol{\phi}_n = \mathbf{k}^{-1} \mathbf{m} \boldsymbol{\phi}_n$)

Therefore:

$$u_{Nn}^{st} = \frac{\Gamma_n}{\omega_n^2} \phi_{Nn} \quad \begin{array}{l} \text{roof (} N^{\text{th}} \text{ floor)} \\ \text{displacement} \\ \text{for } n^{\text{th}} \text{ mode} \end{array}$$

V_b^{st} & u_N^{st} are obtained from a static analysis of the building subjected to (lateral) forces \mathbf{s} .

We define:

$$\bar{V}_{bn} = \frac{V_{bn}^{st}}{V_b^{st}} \quad \& \quad \bar{u}_{Nn} = \frac{u_{Nn}^{st}}{u_N^{st}}$$

TABLE 12.11.1 MODAL AND CUMULATIVE CONTRIBUTION FACTORS

Mode n or Number of Modes, J	Force Distribution, s_a				Force Distribution, s_b			
	Roof Displacement		Base Shear		Roof Displacement		Base Shear	
	\bar{u}_{5n}	$\sum_{n=1}^J \bar{u}_{5n}$	\bar{V}_{bn}	$\sum_{n=1}^J \bar{V}_{bn}$	\bar{u}_{5n}	$\sum_{n=1}^J \bar{u}_{5n}$	\bar{V}_{bn}	$\sum_{n=1}^J \bar{V}_{bn}$
1	0.880	0.880	1.252	1.252	0.792	0.792	1.353	1.353
2	0.087	0.967	-0.362	0.890	0.123	0.915	-0.612	0.741
3	0.024	0.991	0.159	1.048	0.055	0.970	0.431	1.172
4	0.008	0.998	-0.063	0.985	0.024	0.994	-0.242	0.930
5	0.002	1.000	0.015	1.000	0.006	1.000	0.070	1.000

Observations:

- (1) The *modal contribution factors* are larger for base shear than for roof displacement.

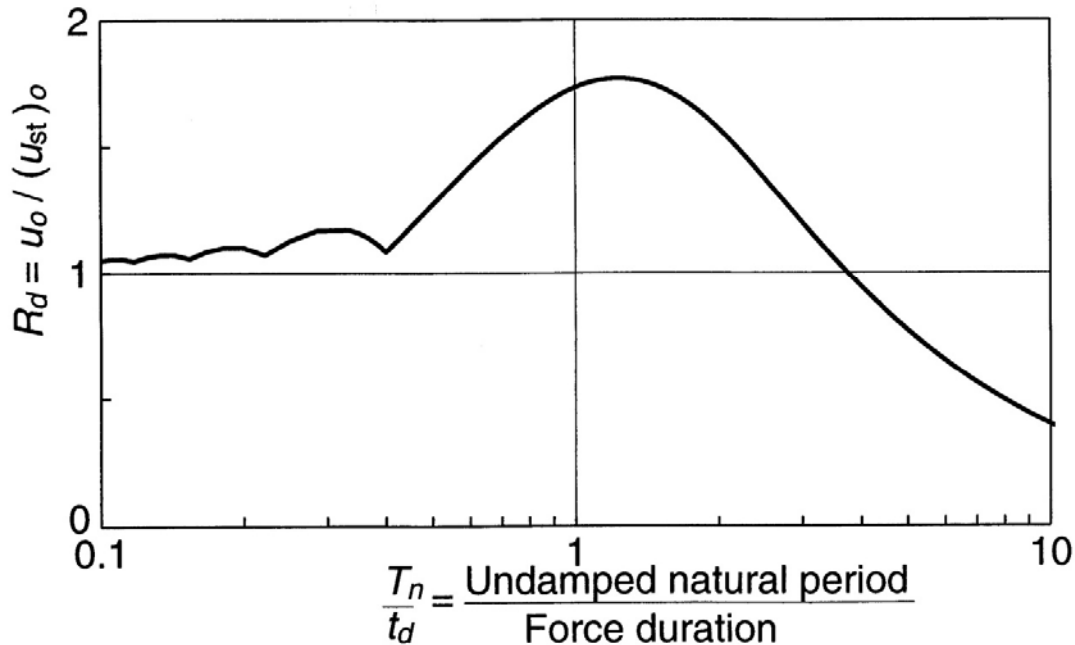
Higher modes contribute more to base shear (and other element forces) than to roof displacement (and other floor displacements)

- (2) The *modal contribution factors* for higher modes are larger for force distribution s_b than for s_a .

Higher modes contribute more to a response in the s_b case.

The modal expansion of s_a & s_b suggests the same conclusion.

Dynamic response factor for half-cycle sine pulse force



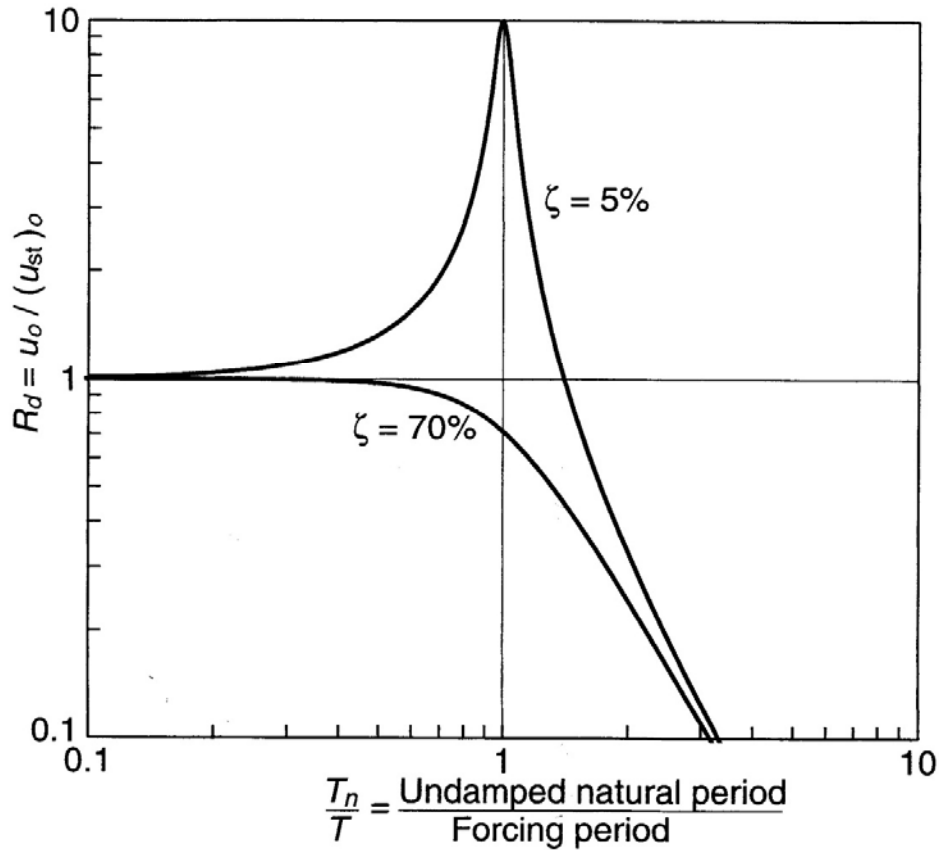
How R_{dn} ($=$ the value of R_d for the n^{th} mode) for a given excitation $p(t)$ varies with n

depends on

where the natural periods T_n fall on the period scale.

In the case of **pulse excitation**, the above FIGURE shows that R_{dn} varies over a **narrow range** for a wide range of T_n and could have **similar values for several modes**.

Dynamic response factor for harmonic force



For **lightly damped systems** subjected to **harmonic excitation** (Ω), R_{dn} is especially large for modes with natural period T_n close to the **forcing period** T , and **perhaps** these are the only modes that need be included in modal analysis

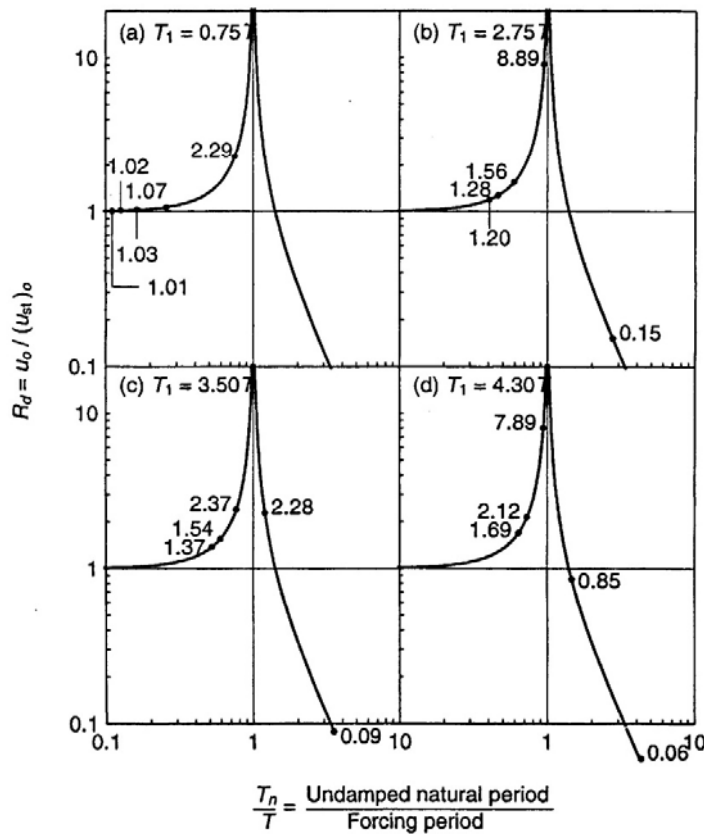
unless the modal contribution factors \bar{r}_n for these modes are much smaller than for some other modes.

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TABLE 12.11.1 MODAL AND CUMULATIVE CONTRIBUTION FACTORS

Mode <i>n</i> or Number of Modes, <i>J</i>	Force Distribution, <i>s_a</i>				Force Distribution, <i>s_b</i>			
	Roof Displacement		Base Shear		Roof Displacement		Base Shear	
	\bar{u}_{5n}	$\sum_{n=1}^J \bar{u}_{5n}$	\bar{V}_{bn}	$\sum_{n=1}^J \bar{V}_{bn}$	\bar{u}_{5n}	$\sum_{n=1}^J \bar{u}_{5n}$	\bar{V}_{bn}	$\sum_{n=1}^J \bar{V}_{bn}$
1	0.880	0.880	-1.252	1.252	0.792	0.792	1.353	1.353
2	0.087	0.967	-0.362	0.890	0.123	0.915	-0.612	0.741
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4	0.008	0.998	-0.063	0.985	0.024	0.994	-0.242	0.930
5	0.002	1.000	0.015	1.000	0.006	1.000	0.070	1.000

Dynamic response factors R_{dn}



In judging the contribution of a natural mode to the dynamic response of a structure, it is necessary to consider the combined effects of the modal contribution factor \bar{r}_n and the dynamic response factor R_{dn} .